

Using the TRIDENT Drill to Assess Geotechnical Properties of Probable Icy Lunar Regolith on Upcoming South Pole Missions

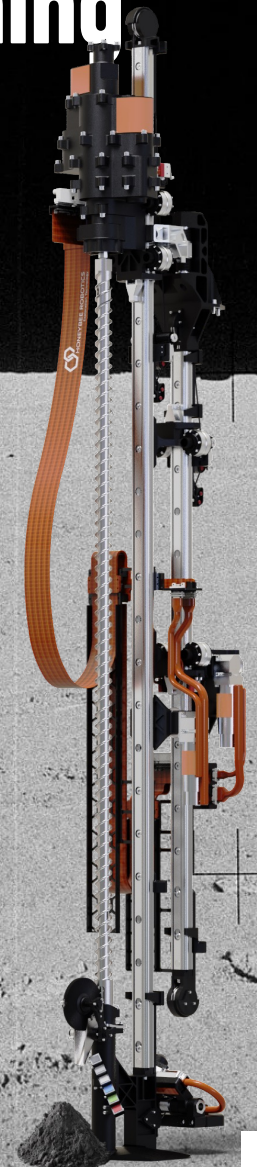


HONEYBEE ROBOTICS



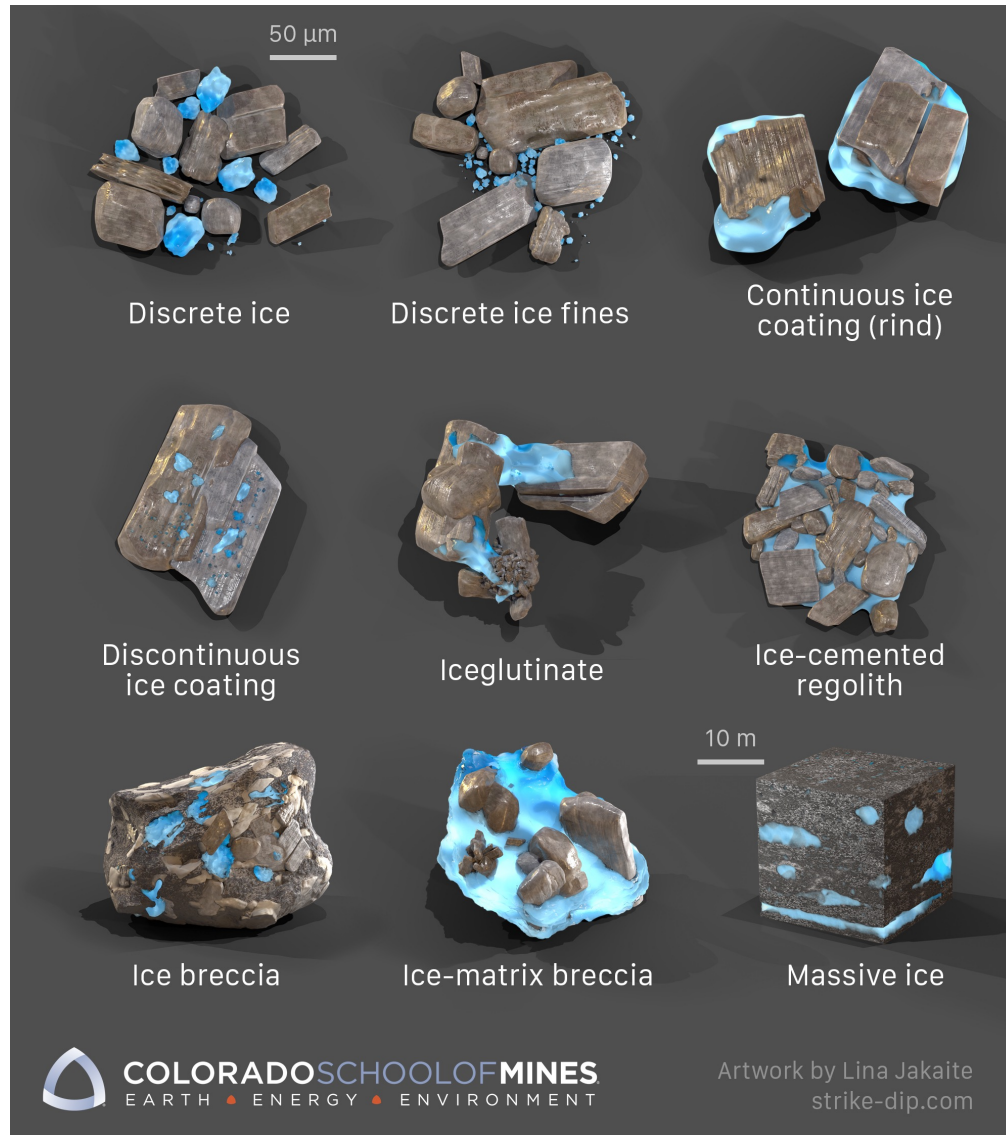
Isabel King

K. M. Cannon, V. T. Bickel, K. Zacny, D. Bergman, S. Goldman,
R. Lin, P. Chu, P. Creekmore, P. Ng, and the TRIDENT team



AS14-68-9407

What We (Don't) Know About Lunar Ice



Remote sensing, lunar volatile transport models, and LCROSS point to water ice deposits in the lunar south pole

Extent, concentration, and texture of the regolith-ice mixture are unknown

This information has major implications for ISRU



PRIME-1 and VIPER Missions



Goal: determine distribution and abundance of lunar polar volatiles

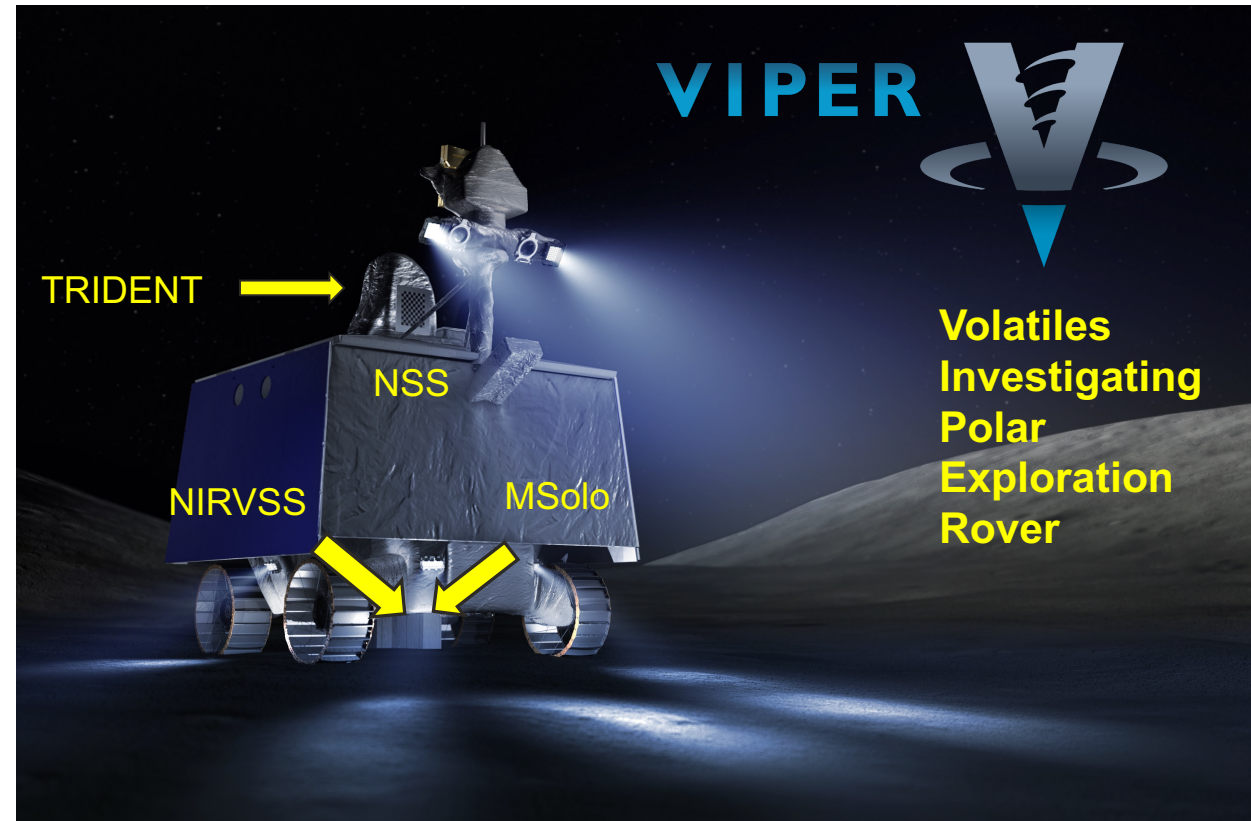
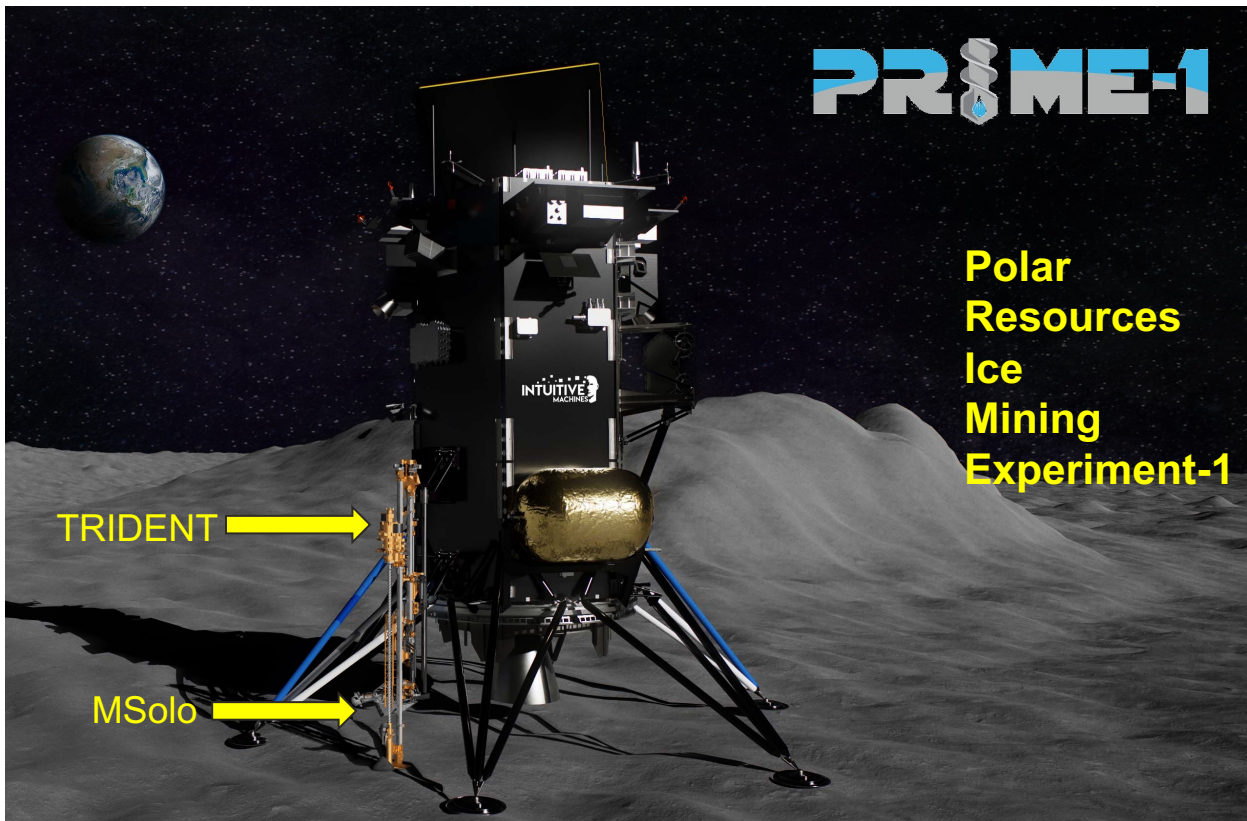
Instrumentation:

Neutron Spectrometer System (NSS) for hydrogen;

Mass Spectrometer Observing Lunar Operations (MSolo) for volatile characterization

Near-Infrared Volatiles Spectrometer System (NIRVSS) for volatile and mineralogical characterization

The Regolith and Ice Drill Exploring New Terrains (TRIDENT) for bringing cuttings from the subsurface ... and more



TRIDENT Overview & ConOps



HONEYBEE ROBOTICS

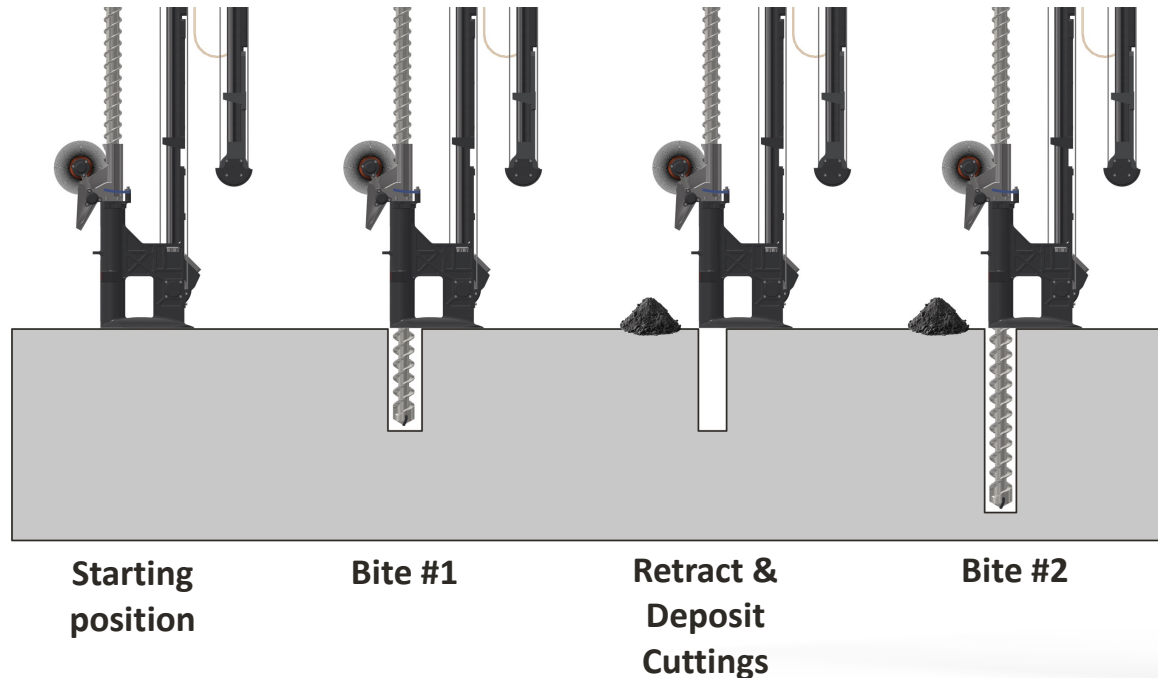
Key Facts:

1-meter rotary percussive drill

Two linear stages: drill head travels up/down feed stage, feed stage travels up/down deployment stage

Footpad at the end of the feed stage deploys to bring the drill to the ground

Bite Sampling



Drill Head
Auger / Percussion

TRIDENT Drill

Deployment Stage

Feed Stage

Auger

Brush / Chute

Cuttings Pile

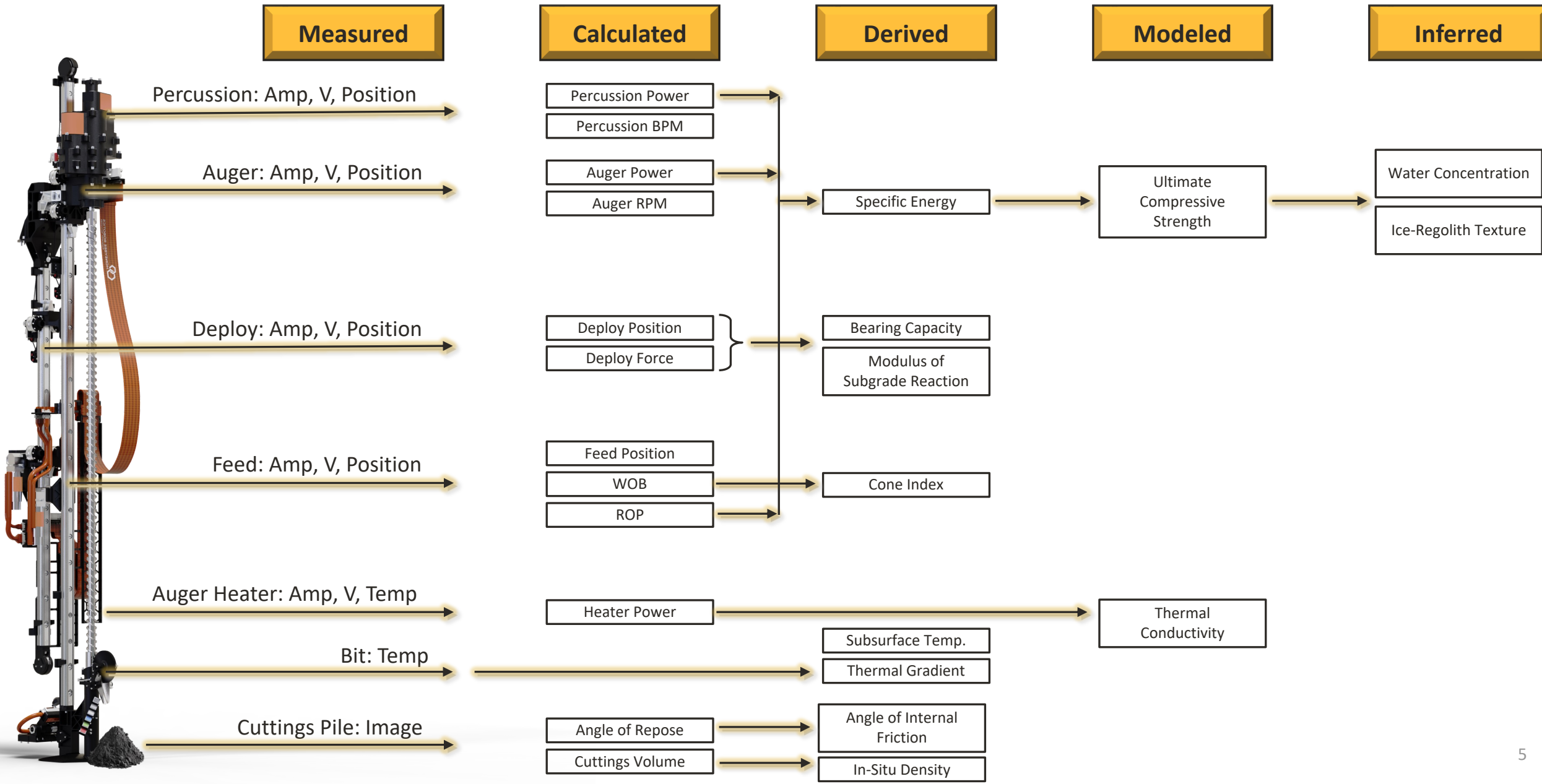
Avionics Box



TRIDENT as an Instrument



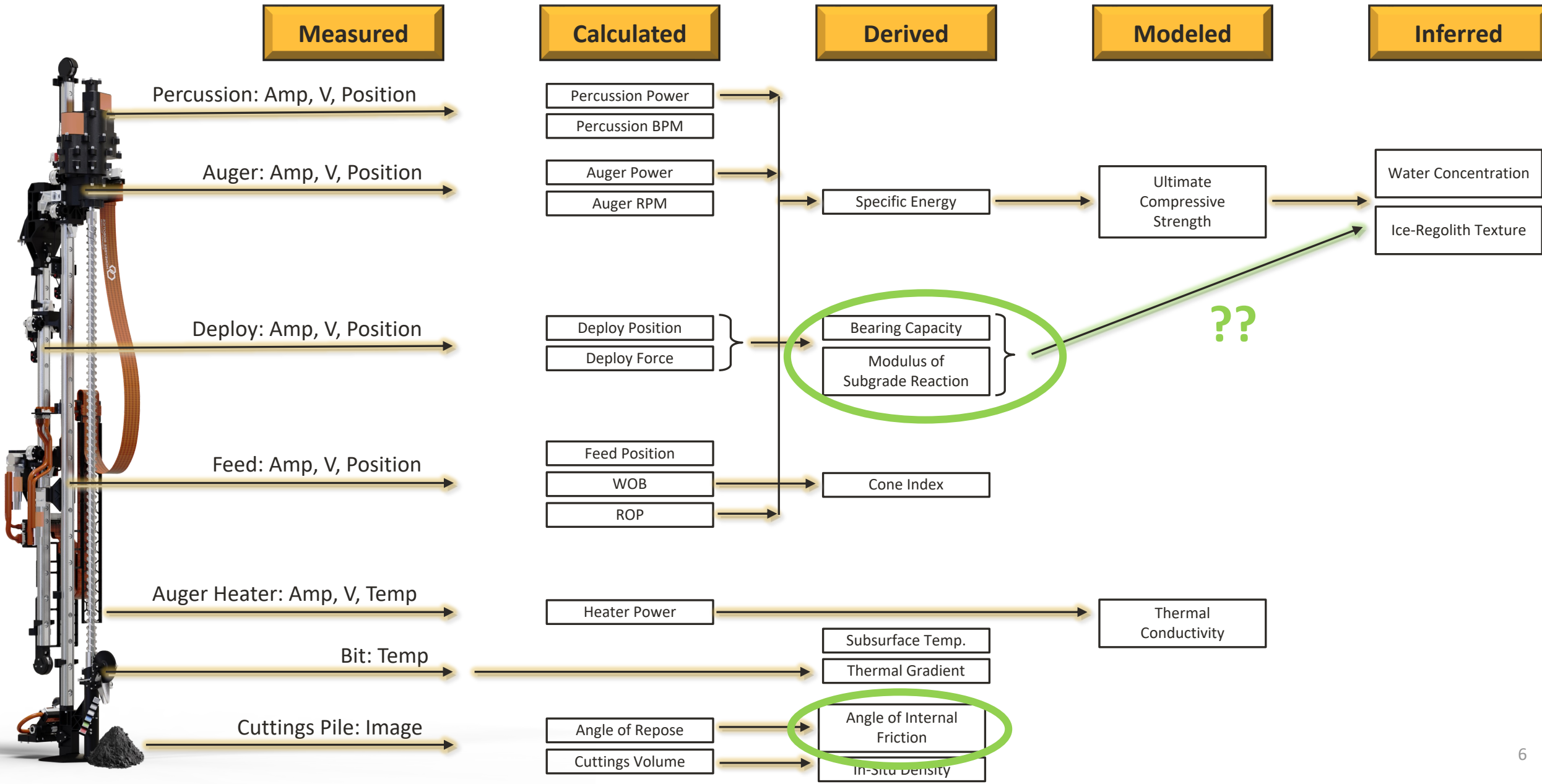
HONEYBEE ROBOTICS



TRIDENT as an Instrument



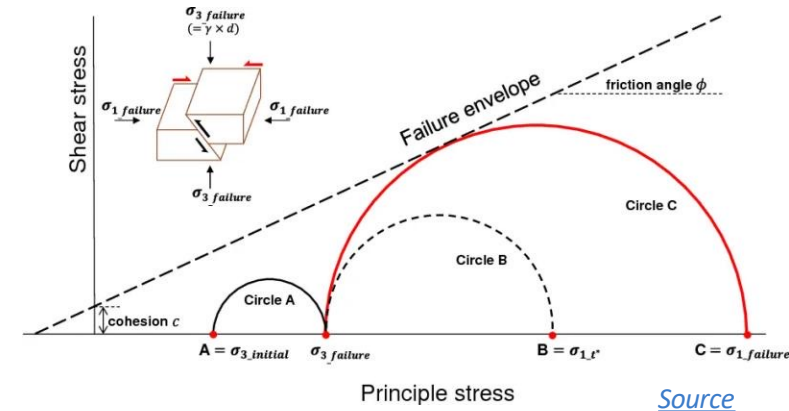
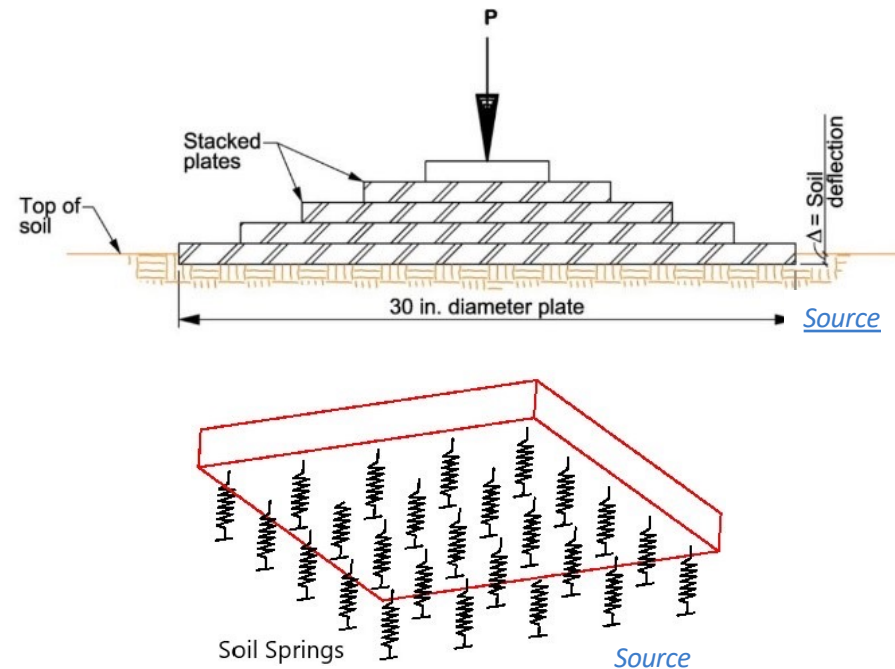
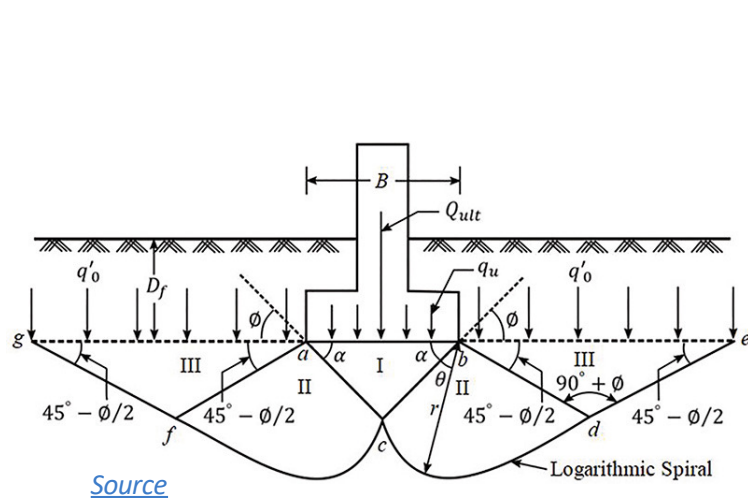
HONEYBEE ROBOTICS



Key Definitions

These properties are routinely measured and used in civil/geotechnical engineering applications on Earth
On the Moon, these are useful in both engineering design and geological interpretation

Geotechnical Property	Explanation
Bearing Capacity	You will sink in a dry granular medium until your bearing pressure (F/A) is equal to the bearing capacity (q) of the soil
Modulus of Subgrade Reaction	An estimated stiffness or spring force of the soil
Angle of Internal Friction	Measure of a soil's ability to withstand shear



Apollo Bootprints



HONEYBEE ROBOTICS

Image
Credit:
NASA



AS11-40-5877



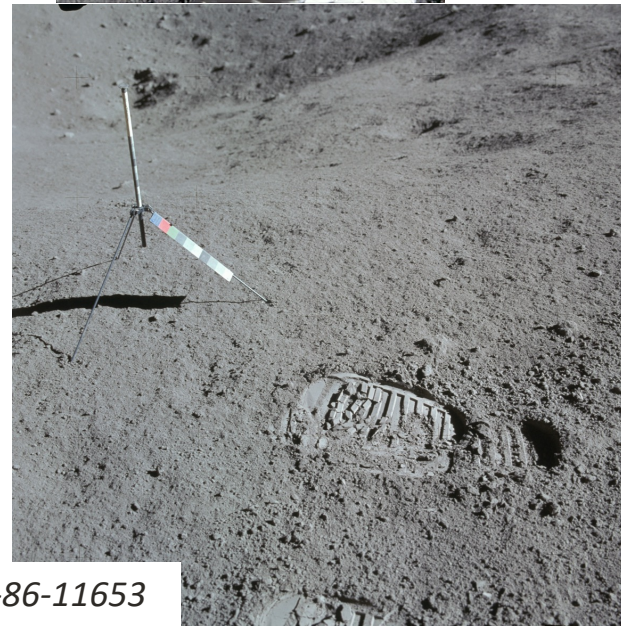
AS11-40-5880



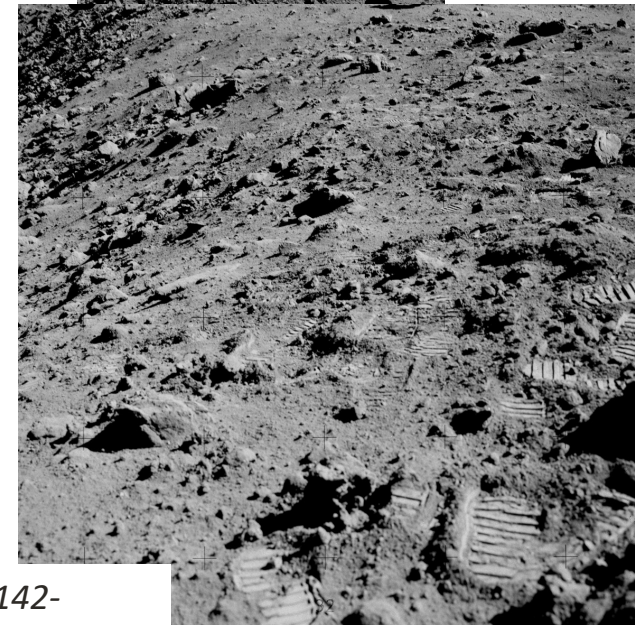
AS12-48-7045



AS12-49-7284



AS15-86-11653



AS17-142-

Expectations in Lunar PSRs

Will geotechnical properties in PSRs be any different than Apollo landing sites?

Competing theories: surface-level ice or lack of thermal cycling could result in differences, but estimates of bearing capacity from boulder tracks look very similar to elsewhere on the Moon

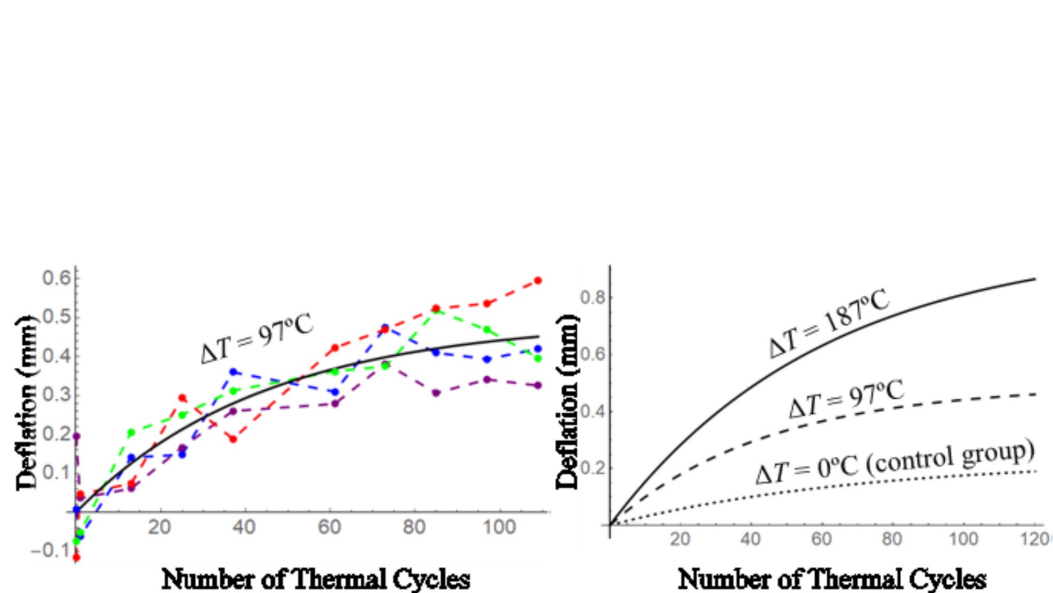
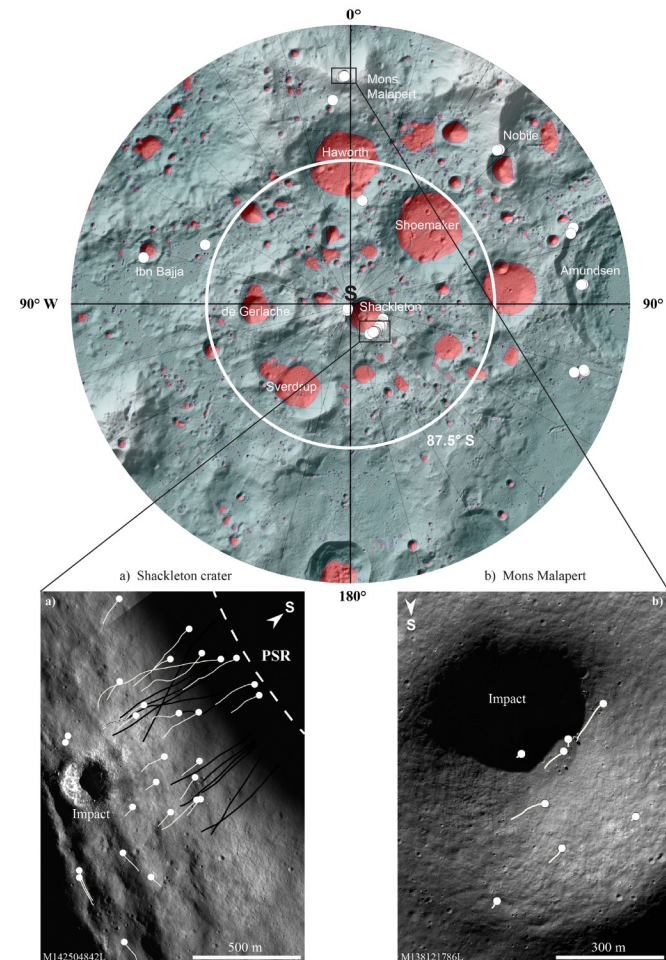
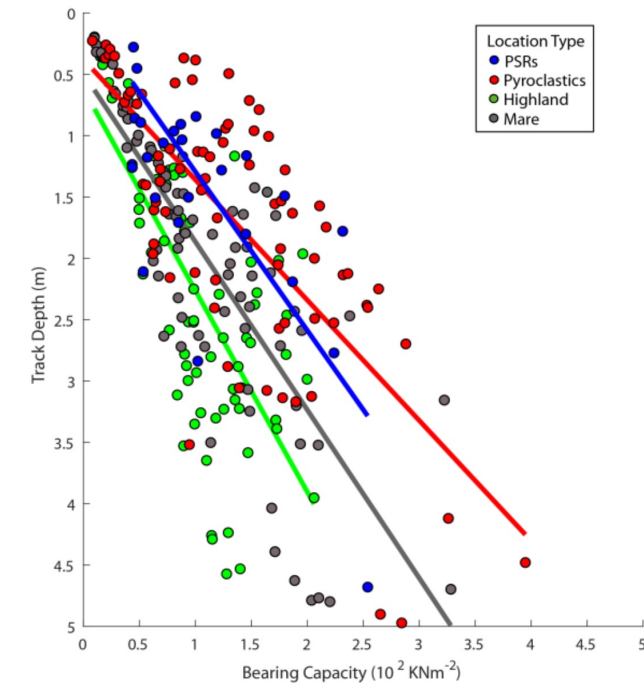


Figure 2. Results of Thermal Cycling. Left: For the $\Delta T = 97^\circ\text{C}$ case, four samples in dashed colors with the fitted function in solid black. Right: the fitted functions for three cases, $\Delta T = 187^\circ\text{C}$, 97°C , and 0°C (the control group that included all sample handling but no oven power).

Source: Metzger et al. (2018)



Source: Bickel et al. (2020)

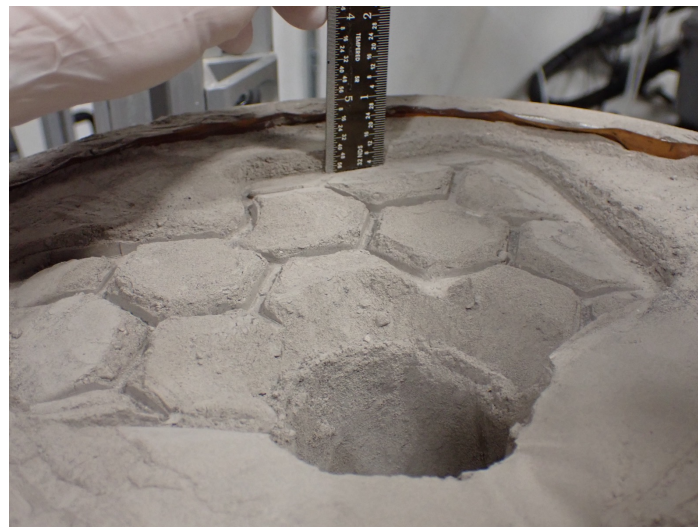
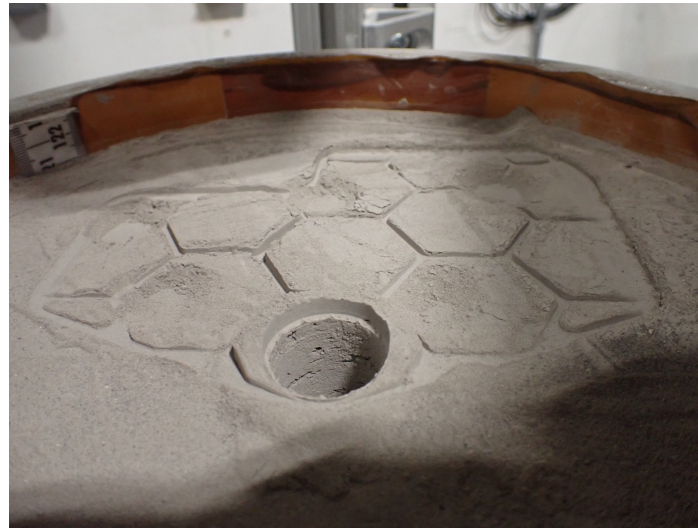


Source: Sargeant et al. (2019)

TRIDENT Footpad Prints

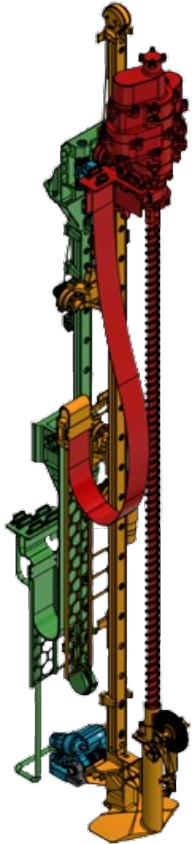
TRIDENT enables more precise measurements: known force and displacement

Possible experiment on VIPER: two-stage footpad deployment



Two-Stage Footpad Deployment

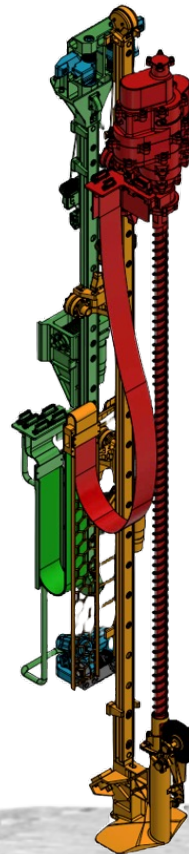
1. Drill stowed on lander/rover



*Yellow rail travels along
green rail to place footpad
on the ground*



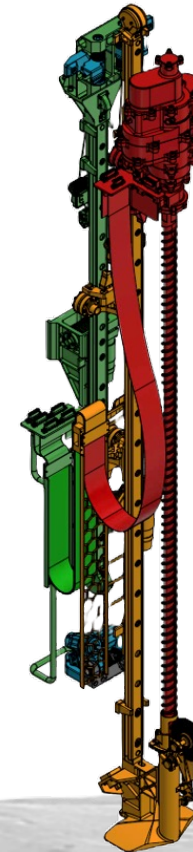
2. Command footpad to deploy to min. preload (47 N)



*Yellow rail travels along green
rail to sink footpad further into
ground, typically +3-5 mm*

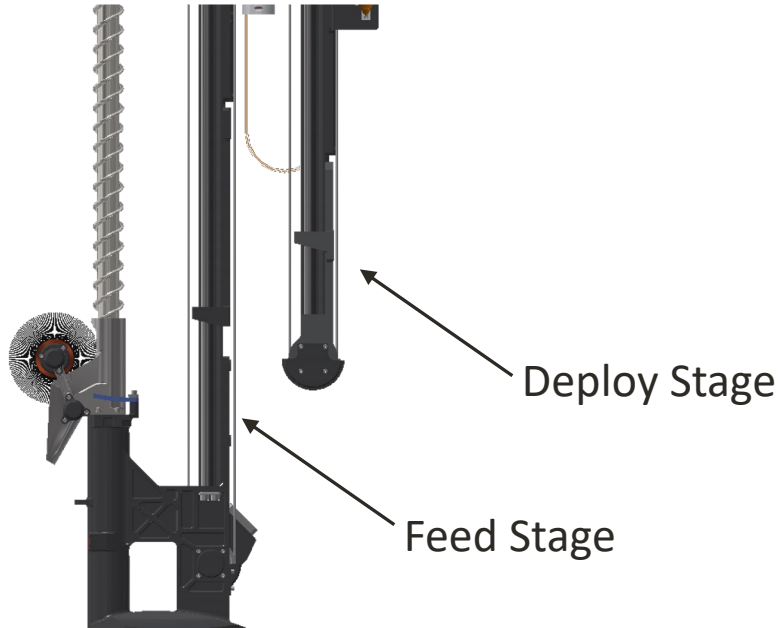


3. Command footpad to deploy to max. preload (183 N)



Two-Stage Footpad Deployment

STOWED



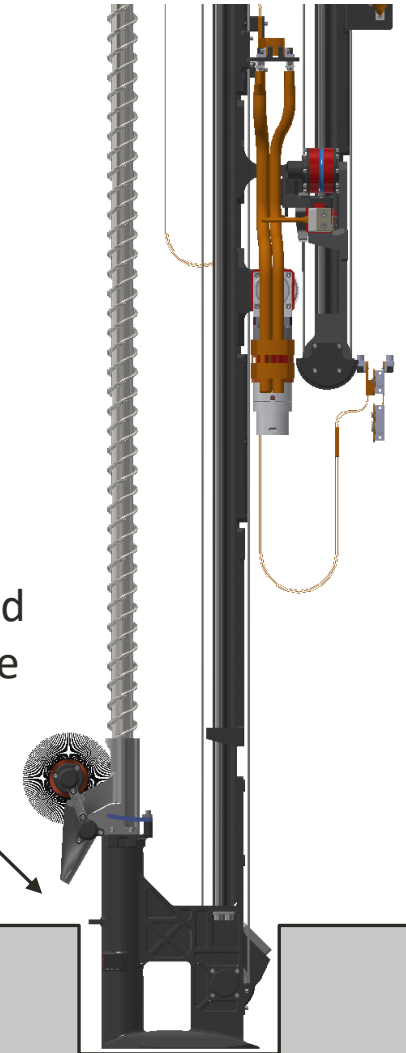
Deploy Stage pushes footpad into ground with 47 N force

MIN. DEPLOY



Deploy Stage pushes footpad into ground with 183 N force (exaggerated)

MAX. DEPLOY



Experimental Results – Dry/Ambient NU-LHT-2M



HONEYBEE ROBOTICS

MODULUS OF SUBGRADE REACTION

Test ID	d (mm)	k (kN/m ² /m)
006B	4.55	2860.8
008B	4.23	3080.9
007	4.72	2762.2
008	3.09	4216.0
009	4.72	2762.2
010	4.39	2966.8
011	3.25	4005.2
012	3.58	3641.0
013	3.41	3814.4
014	4.06	3204.1
015	4.07	3204.1
017	4.88	2670.1
019	4.88	2670.1
020	4.39	2966.8
021	4.55	2860.8
022	3.74	3482.7
024	4.07	3204.1
Avg:	4.15	3198.4

BEARING CAPACITY

$$q = \frac{F}{A}$$

At 4.15 mm depth, our simulant has the following bearing capacity:

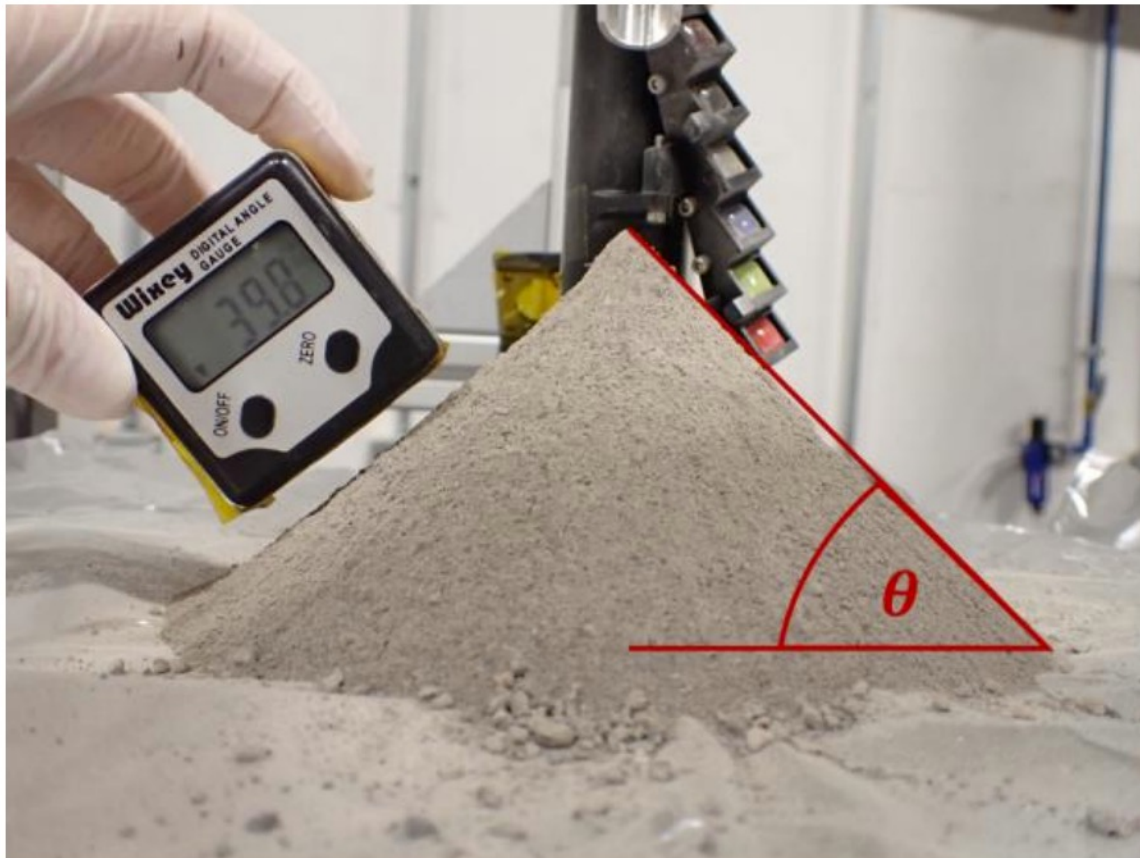
$$q = \frac{183}{140 \text{ cm}^2} = 13.07 \text{ kPa}$$

Angle of Internal Friction

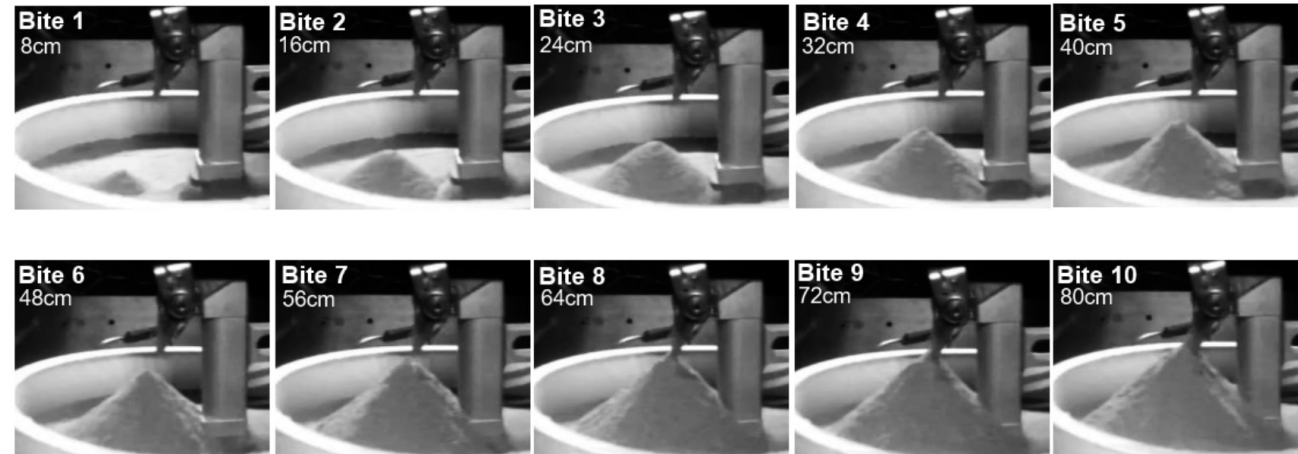
Angle of repose is a good approximation for angle of internal friction, which is a fundamental soil property

Estimation of cuttings pile angle of repose possible from planned images

Apollo regolith estimates were 35-50 degrees



Cuttings pile after 10 bites into NU-LHT-2M at HBR lab



Cuttings pile growth over 10 bites in TVac testing at NASA GRC

Future Work



HONEYBEE ROBOTICS

TRIDENT EU test campaign is ongoing

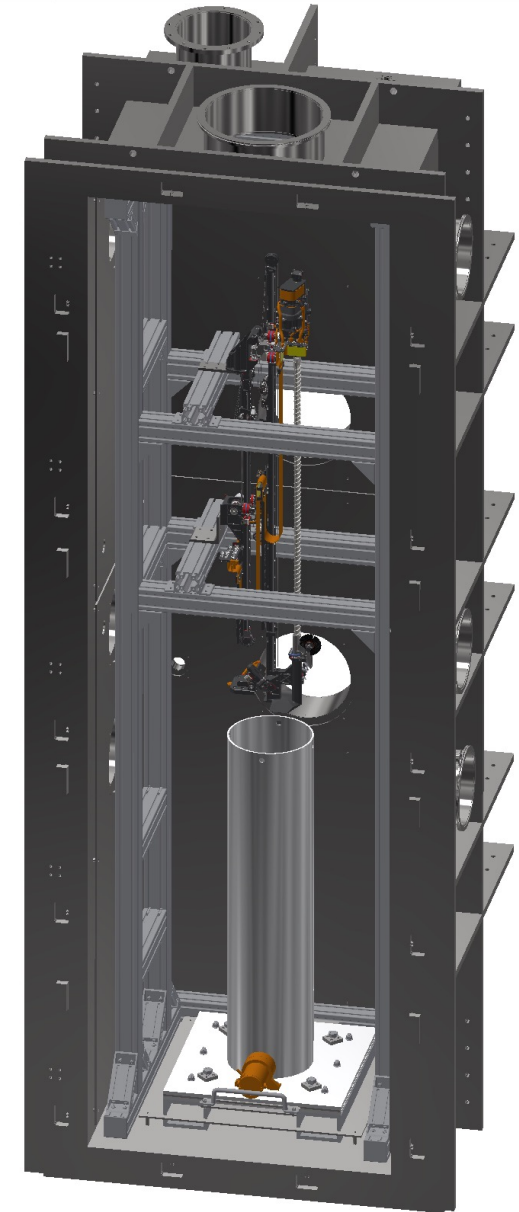
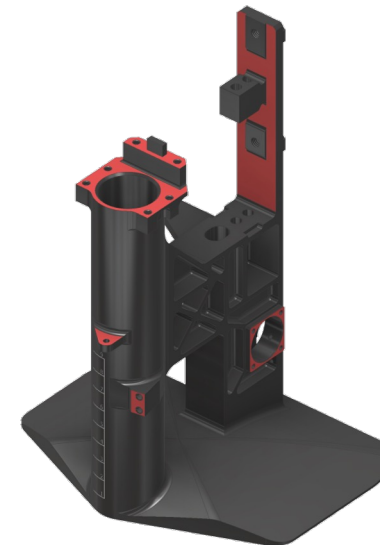
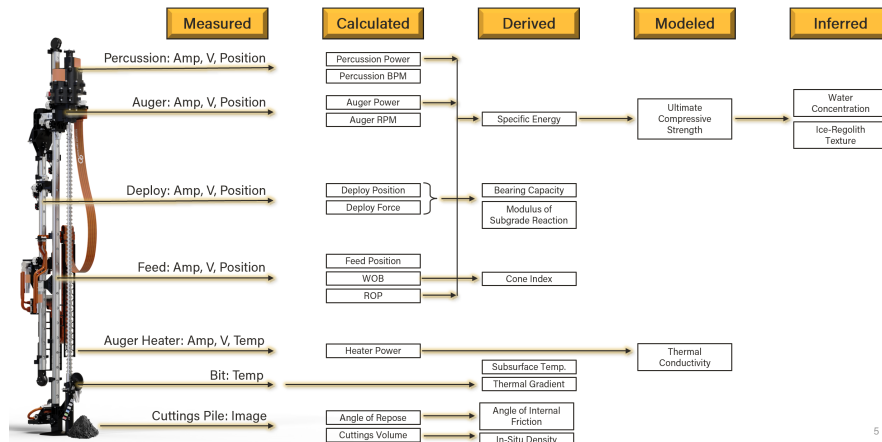
Only ambient testing in NU-LHT-2M thus far

TVac testing in varying texture and concentration ice to come

More cohesive LSP simulant to come

Comparison to other measurements: VIPER's rover team is making similar engineering measurements, ex. wheels for bearing capacity

Make these measurements on the Moon! Provide novel insights into the geotechnical properties of probable icy lunar regolith in PSRs





HONEYBEE ROBOTICS

